



UNIVERSITI TUN HUSSEIN ONN MALAYSIA

**FINAL EXAMINATION
SEMESTER II
SESSION 2010/2011**

COURSE : SATELLITE COMMUNICATION
SYSTEM

COURSE CODE : BEP 4243

PROGRAMME : BEE

EXAMINATION DATE : APRIL / MAY 2011

DURATION : 3 HOURS

INSTRUCTION : ANSWER FIVE (5) QUESTIONS
ONLY

THIS PAPER CONSISTS OF SEVEN (7) PAGES

- Q1** (a) Explain the THREE laws that was derived by Johannes Kepler (1571 – 1630) that are able to show the rotation of satellite around the Earth. Use figures or mathematic models to enhance the definition. (9 marks)
- (b) With the aid of sketch or diagram, intepret the meaning of some of the Keplerian elements that are given below:-
- (i) apogee and perigee, (2 marks)
 - (ii) inclination, (1 mark)
 - (iii) semi-major axis, (1 mark)
 - (iv) right ascension of the ascending node, and (2 marks)
 - (v) argument of perigee (2 marks)
- (c) The eccentricity, e , and the semi-major axis, a , are given as 0.0011501 and 7192.335 km, respectively. Calculate the apogee and perigee heights by assuming that the mean Earth radius is 6371 km . (3 marks)

- Q2** (a) Distinguish the position vector, r and the velocity vector, v which can specify the motion of a satellite in an orbital plane. (4 marks)
- (b) The 2-line elements for satellite NOAA 27 are as follows;

```
NOAA 27
1 28654U 05018A 05154.51654998 .00000093 00000-0 28161-4 0 189
2 28654 98.7443 101.8853 0013815 210.8695 149.1647 14.10848892 1982
```

By referring to Table 1, calculate:-

- (i) epoch day in Julian Day, (2 marks)
- (ii) time in Julian century with JD_{ref} is reference time of January 0.5, 1900 = 2415020 Julian days and Julian Century (JC) = 36525 days, (2 marks)
- (iii) Greenwich Sidereal Time, GST, where it is given as,

$$GST = 99.6910 + 36,000.7689 \times T + 0.0004 \times T^2 + UT^{\circ}$$

(3 marks)

- (iv) semi-major axis, a , and (4 marks)
- (v) true anomaly, v , where it is given as below.

$$v \cong M + 2e \sin M + \frac{5}{4} e^2 \sin 2M \quad (2 \text{ marks})$$

- (c) Differentiate between high earth orbiting (HEO), medium earth orbiting (MEO) and low earth orbiting (LEO). (3 marks)

- Q3** (a) For a point rain rate of 11.3 mm/h, the rain attenuation exceeded 0.001% of the time in any year. The altitude of the Earth station and the antenna elevation angle at the Earth station is given as 243.4 m and 21.6 °, respectively. For a satellite communication with a carrier frequency of 15 GHz and the height of rain of 1.95 km, estimate:

- (i) vertical polarization, (4 marks)

- (ii) horizontal polarization and (3 marks)

- (iii) circular polarization. (5 marks)

- (b) By using the rain attenuation for the circular polarization from Q3 (a)(iii) above, evaluate the cross-polar discrimination, XPD, in unit dB. Given;

$$V = \begin{cases} 20 & \text{for } 8 \leq f \leq 15 \text{ GHz} \\ 23 & \text{for } 15 < f \leq 35 \text{ GHz} \end{cases}$$

$$U = 30 \log f - 10 \log (0.5 - 0.4697 \cos 4\tau) - 40 \log (\cos \theta)$$

(8 marks)

- Q4** (a) Depolarization is an effect of orthogonal component that may be generated from the transmitted polarization. Differentiate in between Ionospheric, Rain and Ice depolarizations.

(6 marks)

- (b) A geostationary satellite is stationed at 10° E and transmits a vertically polarized wave. Estimate the angle of polarization at an Earth station at latitude of 45° N and longitude 5° E. Given $a_{\text{GSO}} = 42164$ km and the mean radius of spherical Earth, $R = 6371$ km.

(14 marks)

- Q5** (a) Create a satellite circuit which consists both the B_1 and B_2 modes of interference. Explain the meaning of -3dB contour and coordination, in terms of interference. (5 marks)
- (b) Design a satellite circuit block diagram, which can be used to define the transmission gain factor. (5 marks)
- (c) Given that $L_U = 200$ dB, $L_D = 196$ dB, $G_E = G'_E = 25$ dB, $G_S = G'_S = 9$ dB, $G_{TE} = G_{RE} = 48$ dB, $G_{RS} = G_{TS} = 19$ dB, $U_S = U'_S = 1$ μ J, and $U'_E = 10$ μ J. Given $[k] = -228.6$ dB. Calculate:
- (i) transmission gain, γ , (4 marks)
- (ii) the equivalent temperature rise overall. (6 marks)
- Q6** (a) Global Positioning Satellites (GPS) is a system that can be used for so many applications such as for Military, Navigation, Aviation, Asset Tracking, Surveying and Agriculture.
- (i) Interpret the statement above by giving brief explanation on the usage of the GPS system in Military, Navigation and Aviation. (9 marks)
- (ii) In terms of accuracy, ionosphere does give positioning error in Navigation. Propose 3 methods that can be implemented in order to mitigate the ionospheric error and to improve the performance of the GPS system in Navigation. (6 marks)
- (b) During raining time at tropical countries such as Malaysia, the TV signals that are being broadcasted using the geostationary satellites will be distorted. They are few reasons for the occurrence of this problem. Name and explain TWO (2) main reasons for this distortion to take place. (5 marks)

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**Table 1: Julian Dates at the beginning of each year (Jan 0.0 UT)
 for the years 1999 - 2010**

Year	Julian date (days)
	2 450 000.+
1999	1 178.5
2000	1 543.5
2001	1 909.5
2002	2 274.5
2003	2 639.5
2004	3 004.5
2005	3 370.5
2006	3 735.5
2007	4 100.5
2008	4 465.5
2009	4 831.5
2010	5 196.5

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Table 2 : Reduction Factors

For p = 0.001%	$r_{0.001} = \frac{10}{10 + L_G}$
For p = 0.01%	$r_{0.01} = \frac{90}{90 + 4L_G}$
For p = 0.1%	$r_{0.1} = \frac{180}{180 + L_G}$
For p = 1%	$r_1 = 1$

Table 3 : Specific Attenuation Coefficients

Frequency, GHz	a_h	a_v	b_h	b_v
8	0.00454	0.00395	1.327	1.31
10	0.0101	0.00887	1.276	1.264
12	0.0188	0.0168	1.217	1.2
15	0.0367	0.0335	1.154	1.128
20	0.0751	0.0691	1.099	1.065
25	0.124	0.113	1.061	1.03
30	0.187	0.167	1.021	1

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Constant values and formulas:

$$\mu = 3.986005 \times 10^{14} \text{ m}^3 \text{ sec}^{-2} \text{ (Gravitational parameter)}$$

$$R = 6371 \text{ km (mean Earth radius)}$$

$$K_1 = 66063.1704 \text{ km}^2 \text{ (Earth's constant)}$$

$$\text{Boltzman's constant, } k = 228.6 \text{ dB}$$

$$\mathbf{r} = (r \cos v) \mathbf{P} + (r \sin v) \mathbf{Q}$$

$$\begin{bmatrix} r_I \\ r_J \\ r_K \end{bmatrix} = \bar{R} \begin{bmatrix} r_P \\ r_Q \end{bmatrix}$$

$$\bar{R} = \begin{bmatrix} (\cos \Omega \cos \omega - \sin \Omega \sin \omega \cos i) & (-\cos \Omega \sin \omega - \sin \Omega \cos \omega \cos i) \\ (\sin \Omega \cos \omega + \cos \Omega \sin \omega \cos i) & (-\sin \Omega \sin \omega + \cos \Omega \cos \omega \cos i) \\ (\sin \omega \sin i) & (\cos \omega \sin i) \end{bmatrix}$$